Ehricke, March 11, Seal Beach

Cross-currents ad So much has happen and a so many A but, Let me ask you this Is there anybody to review this? You know as we go on in questions and and answers two things can happen. One can be just a misplacement in time of in connection that one might want to correct me on. What the memory stood by the question one think. The other is one simple forgot something and say O, god I should have mentioned that. Well you could get a hold of me through Ralph. He'll know where I am. I'm at the University of Ala. Huntsville. O.K. Dates aren't very important and when I teach I do a lot of teaching and when I teach students I'm not interested in the dates that things happen. I'm interested in some kind of order. Because its a sequence of events. But I don't care wheather they say it was 61 or 64. Because I'm figuring that out in overall cronology. So I'm more interested in your impressions and interpretations and I'm interested in you opening things to me. Which I can then go to the documentation and check. Cause you know Marshall saved enough stuff to sink this building into must be the sand so far as documentation is concerned. I can't imagine, unbelievable. I MANIN can read all that. So I've got to start at the top with. Can anybody -ead all that? Will ever anybody? I can't imagine unbelievable. It's one of the paradoxes in 20th century history. There's so its and embarrisment of Miches you have so much to do. How do you select what you're goming to look at? And to this it wont be typed out and used verbatum at all. But the material if you are quoted at all you will have a chance to see it later. Our PR people will me or somebody will get a chance to read the whole thing and comment on it and we'll send copies to everybody that is involved in it. So you will have a chance to...

O.K. So feel a little free now any time you want I'll turn it off. O.K. Fine. So give a little more freedom there. O. K. Well, do you have any particular question? Work in the square. Oh. you got to use an hour you an hour you just and lockat it. Well let me say that I personally have always taken the broudest possible approach to space flight. Space flight has always been two things to me, large vehicle and spacecraft. They really have never completely separated and it has always meant to me exploration and application. I never separated these two things. My first book back in 59 I made that point very strongly and I pleaded for more emphasis on practical applications along with oneday the exploration or we'll be stopped in the exploration unfortunately some of this has happened now. We're turning around but it is so some of my thoughts will be dictated a little bit by the kind of frame of reference. In other works I always have tried to think in very large indicative blocks rather than individual thinks so when you see the trend I just want point that out. Now, as you know really we got our feet wet for the first time in modern history with a real crosseyed space craft. If you precussively speaking in the sense that it had the kind of reliability that you need for repeated launchings rather than the earlier test however ingenius they were they had at best a 50-50 chance. This one for the first time was one which in a very rapid sequence was supposed to become a bura company system that at any time could be launapeelbut not excessively trained cool beings all by people or special inventors. Number one number two for the first time an extensive amount of automatic of kinds of control equipment was involved which was one almost completely absent in the earlier thenop

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which was almost completely absent in all the earlier things. The earlier things were essentially flying propulsion systems. That was about it. By and large which may be a little bit of a generalization they had very premative direction devices but for the first this had a full flesh kind of automatic guidance control system even though part of it was on the ground, but it was accurately corrected It's designed very interestingly enough the early design of the E-2 attempted light weight design. But it was always a self-supporting structure. I do remember that in early 42 when we fueled one we two as you know it had fends. There is one of them, the fends were facturand. Actually, so that looked awfully these days dangerous especially we really not a very good idea how far we could go. You know but today we do things we would not have dared to do 30 years ago, 40 years ago. And so the statue was a little bit reinforced. And fundamentally the E-2 emerged as lightweight by the then existing standards. As a very massive eagle by some of todays standards standards not all. Now this line of safe prencipal design safe and relatively massive design is perhaps the preneiple trademark of the quote, unquote "the German team," I'm an exception because I did not Wernher do that on center. That had certain and discuss it with Wuner TON Drawn . But you look at Jupiter, look at the Saturn, serious VonBoun sturdely its very stordaly built. An the American wine if you want to call it that even though some of us were in on this one too. Started off a little bit differently Already Prof. Overt in his book and its really difficult to do something that Overt hasn't . Overt in his book suggested that because he was very much aware Overt was a German \_\_\_\_\_. He was very much

aware of the problems of the mass ratio. And Physics Proffessors and other people in those days gave him a heck of a time in the 20's late 20's and early 30's that you could never build something light enough to come up the that would be sufficient to send something in orbit. And he felt it could be done, the suchellentry if we improve our steel so that we could make it so thin that the pressure stabilizer he called it balloon contruction and the American word is more in technilogical Pressure Stabilizer. Now independent of Overt this thought was devloped by Dr. Bosserk, at Convia. When he worked on the MX774 the Atlas vehicle ulacle The Atlas was indeed based on this kind of structure. I talked to Dr. Bosserk with whom I worked very closely my good friend it bothered and killed him a little bit but he honestly did not know this. This was strictly his own genius that came up with this idea and saw that as a giant convayer I liked it very much I thought it was a good way to go. But for the very large eagles it was not found to be too acceptable, there were just too many forms involved. And fierce apprehensions of a very difficult structural situation. Especially if you have one of the state's vehicles although I was completely convences that it could be done on a two stage vehicle of modest demensions and that's what became CENTJUR Centann Centile. Now Centile you see what I wanted there at least why I actually deviated from the heavier design infact actually was even a little bit bolder still than the Atlas was. Was not really because I like that kind of frame and ship to much as more due to the fact that the smaller you become the lighter must you build to maintain a certain mass ratio. Now the oxygen-hydrogen propellent is terrific because of its high specific impulse. But its very it can be very

easily degraded. If your mass ratio isn't good. And that means therefore attitude especially in smaller type vehicles the oxygen hydrogen can practically be wiped out / If you are not very careful therefore very small rockets are just not suitable for the use of oxygen, hydrogen. I was convenced to the point that Joime it became really almost a crusade that we had to move rather quickly to hike specific impulse high energy propellents because that would be the only way in which we could make a real breakthrough toward space flight. With our vehicles, and while the airforce investigated a large number of advancements including florine and hydrocene other things along these lines. I remember these things very vividly in the late 50's. People were surrounded by alternatives. As usual there was no concensis to be achieved, what to do. I personally did continue to push for oxygen-hydrogen for a very simple reason. Do.1 Oxygen technology we had. We had it from the V-2 meantime on the Atlas we Thor had it on Jupiter and we had it on the four it was well developed technology. Now, so therefore only one new fuel had to be gotten. The more we went to technology florine in the new fuel we had to develop also florine also for all the poisons and the messey thing that goes with it. And it wasn't really necessary because when you ware an oxygen \_\_\_\_\_ or if you want, and went to oxygen-hydrogen you make already 92 to 93 precent of the junk that you possibly make even if you would go to flurine-hydrogen. So for the last five or seven percent to go through a completely new oxidizer regardless of what its advantages of density and all this were seemed to make very little sense. To go to a new fuel to a new oxidizer but keep the same fuel didn't give you much of a performance

increase. The second reason, why I went to hydrogen rather than hydrocene are anything like this was simply because I was convenced from my nachef

that the time of the nuclear propelled spacecraft would not be far away. Perhaps a nuclear propelled launch vehicle

in terms of that and the only rocket fuel that made sense with a nuclear heat exchanger rockets is hydrogen. Your special impluse deteriorates very rapidly the moment you go away from hydrogen go to hydrogen or anything like that or ammonia. So it wasn't very worthwhile to try anything else but hydrogen. So it seemed to me it was so logical to use the hydrogen that I could never really uerer be interested in any other solution. On top of it there were obvious advantages at every link and while these things were held very classified one couldn't talk much about it. Those of us familiar with jet propulsion could very easily deduce that if we were to replace gasoline by hydrogen that we could build aircrafts of fantastic performance companitively speaking. Everything seemed to me in space craft or air craft ultimately to go to hydrogen. But hydrogen a nasty stuff, a difficult stuff to handle very cold, very illusive, very evaporated and in small it had that problem. I didn't believe that we could justify the development of a completely new large rocket vehicle. Based totally on oxygenhydrogen, people who are not quite convenced that hydrogen would work and the best way to break the door down to begin to at least try to open it anyway. Was to select a vehicle that would not be to big a development process like the Atlas or even something bigger into multi-billion dollar program. But something smaller and at the same time one that would fit on the existing boaster and so I

started to look at this Atlas vehicle so if we straighten this out ten-foot diameter its not the ideal size on the height oxygen vehicle but it just might just be large enough so I can get away with a and solve the massfraction and mass ratio problem if I am using pressurized structure and maybe drive a little bit further and so I felt its imphasis booster and launch vehicle a size that was reasonable to break into and experience with pressurized structure that was available at convey it seemed to fit off. And so I went that route and that route was of course quite adverchent from Huntsville route. We had many frenge discussions many things that in those days that we argued about and Huntsvillen themselves looked at their larger launch vehicles that they were taking to investigating also at the possible ility of a floor way structure. It had in the meantime come to the point that in those lets say in the late 50's and the early 60's really I think that the two top teams in the country at that time were the Marshall team and our Convay team. The Julian team began to come up and wasn't really a bowling are any other major rocket team. And the Huntsville team went the heavy structural route and the Convey team was proud of the light structural route they went that route. We many of our proposals were based on pressurized structure for other stages of saturn later on / and Huntsville had great trouble to accept it and eventually decided not to accept it fundamentally so that the action went into heavy. So that's one important difference perhaps that characterized the whole situation now the pressurized structure has proven itself fabulously. And if you look at the metamorphasis that the saturn vehicle has gone through compare to what the somewhat malined centour had gone through you will find

that the centeur even today as it strives works virtually the same as my department and my people and I had another design on the drawing board back in 59. When we proposed very little change the Saturn vehicle had a different instrument. Initially, Marshall worked at very large booster vehicles. Completely new boosters then I think it was a suggestion of our part of I remember correctly that alpha say well look you have all those pretty nice sized \_\_\_\_\_\_?

cluster those together. They had also looked at clusters but different types of clusters but most of them that I saw at that were almost more like Satern five type boosters although they weren't in that size. They were more in the Saturn IB size range but they were sfort type \_\_\_\_\_ stages - new stages and so forth. That cought on at Marshall and they really went into this and said yes that looks really terrific we can actually build a very nice and dafty booster out of these out of such and arrangement of Jupiter and rinestones around it. CNCOPE But then there was the question of what do we do for an on cour now in the upper stages and oh we had all wild things I use to tell Von Braun that this whatever you're doing its not going to work its not going to give very good performance unless its sentour tipped. And so we had that kidding with Von Braun sentour tipped Saturn family he said I'm not sure wheather that flimsey little thing up there is going to work. You see you need an Oxygen-hydrogen stage you should really construct an oxygen-hydrogen stage up there but I don't mean we have such a big development program we really shouldn't yet. Was he worried about handling hydrogen? Yah, somewhat. Somewhat but not too much I remember that he asked me it was in 47 or so that somebody had published an atricle I remember

somebody had published an atricle at that time I think in the <u>Journal of American</u>
<u>Rocket Society</u> I forgot at the moment the name Anyway in which he proved that
Oxygen-hydrogen was an excellent propellent and I know that Wuyer was unhappy about it and he says I don't think this is right. I want you to investigate and
work out a comparative manathedelity for comparitive analysis of rocket propellents X
Now the point was we weren't always necessarily that accurate in those stages
When these articles or when these arguments came up somebody might have
meant for a spacecraft and talked also about the logic \_\_\_\_\_\_ somebody else
\_\_\_\_\_\_ well I don't agree with you. And so different \_\_\_\_\_\_ as possibly if so
so I set down and wrote this out it was later on published in the <u>Journal of</u>
American Rocket Society and the results basically really was

the results really was indeed for centain conditions heavier propellents were better and other lighter propellents were better. O.K. on that we were kind of settled. Well I used to say Wurner is and this in friendship but he is fundamentally almost more a concervative, the inavator. And for the long time for him to really accept the thought of fuclear powered hydrogen space craft. Remember in his master he made a point Using dense propellent I remember that I was a little bit hesitant I worked the propulsion system of on hi s staff at that time for this March Project that he had and he wanted bevery much to have dense briffients. He kept on saying that its probably safer and if we get a hole puncture in the way mars in the hydrogen tank then everything is down the tube and so on and so forth. And so we settled at that time if I remember correctly on wetfuming nitric acid and hydrocene something like that. And I gave the engines

high pressure and I went to hydrocene to get some decent spacific impulse out of this and the thing got completely out of hand, because the desire for his propollant use is so much. In anycase it was in those days WERNNER that the situation started and Wuerner he was fully aware of the higher performance and he liked it but he felt that maybe the time wasn't come yet. So his is presend were all based dense and propellents. Very relunctantly he once looked at Atlas as a second stage as a second stage which seemed to be stordier and met more his condistructure tions of non pressurized statue and so far I kept on at it we had many meetings aure in those and our friendship kept on all the same he needed a needed a centour tipped Saturn. O. K. now low and behold along came this I still have some I remember this of an unbelieveble variety of the history and you know \_\_\_\_\_\_ even better than I do and probably have a document 1 . The initial C-1 and C-2 and C-3's is just mourderously different from the final ones, so they went through trememdous changes, at that time and it wasn't really settled .-Until a committee was established. I think it was a joint NASA DOD Committee which as a result of our relatively good progress that was still before the first on Centaul way the flight but the way we went ahead and designed it and the engine work begin to impress people. My god, maybe it isn't so bad after all. It wasn't a donment Silverstein! Silveestein The silver stand committee. No. not silver stand. It was Caveno somebody like that and it was a joint DOD NASAcommittee . Have a committee Silveestein I don't its possible that silverstand was on but he was not the name carrying chairman. Their recommendation was forget about anything other oxygen hydrogen. And

verified what basically was not so difficult \_\_\_\_\_. My God, if you go through trouble to build such big a vehicle practically defeating your purpose if you don't come up with oxygen-hydrogen ? O.K. that desiding that then to really the first Saturn one where they well look you have now already stage you have the \_\_\_\_\_\_ engines and the search Why not take the centour. So then you have a makeshift second stage with was it 8 or 10 or so that with the engines on it or 12, six, six \_\_\_\_\_\_ report they couldn't get \_\_\_\_\_ uprated so they Contaur went back to the six. With the six and which was kind of a super centile rating Centaur very much design after that and then the centile on top. Douglas build that did They visited they get information from \_\_\_\_\_. Yes they got information from us. How much would you could you guess at I think practically everything they want to know they could get there was not that kind of a competition. It wasn't we developed it from government money so we felt it was government property. CENTUR and NASA should profit from it. Anyway in the meantime centile which had a terrible history of top level management was kicked around between Headquarters Airforce and NASA and . And NASA was taking this over while NASA headquarters was in its formative stages everybody has their own idea about what should be done which is unbelievable. And you couldn't get any answer when you tried to get a decision, about fundings. Well, Von Braun was manag-Centaur ing Senter for awhile. Not until very later. Then it went up to Louise? Yeah. ?allas started out with Alpha and they couldn't 2 \_ and they couldn't \_\_\_\_\_ and then it went to the air 9

force and then the airforce lost interest or they had other problems and then it

tree so I got along famously with Haunze and it was a really good time. But still the proper management structure was not there and it wasn't Marshalls' baby and Marshall is too great of a team to be able to absorb somebody elses creation I was very difficult. Marshall is not a processor, Marshall is a creative team. You see and creative people have difficulties processing other peoples material. There is always a conflict. In the you see that why for example Marshall got along so famously on the first stage of Saturn the S-Fwith Boeing because Boeing which was desperate in those days for business went all out and pard we are here to please you and my dear girlfriend Rojeck just absolutely loved that whereas he got from us often \_\_\_\_\_. Yeah \_\_\_\_? there was one famous statement which I was not was in the room when it was made I better don't say that even on the was felt it was very knowledgeable and there was of couse \_\_\_\_\_ we have our opinion and you have your opinion and thats always very difficult you should have creative teams. But don't force them on each other. Your technilogical development of a nation is big and as active as the United States has enough room to let an independent dealer, now Boeing in those days took a different teek at the whole thing. Now we want their business Wernhee and we know how to get it. We go to Wuerner we go to \_\_\_\_\_ and say how do you want us to do it? \_\_\_\_\_\_ I give you all the drawings and before I understand before they change crew they went back and have asaid

it worked famously and it is probably good that it worked this way. Because if the Saturn would have been build betweens teams that were <u>conflict</u> with each

\_\_\_\_\_\_\_there was some feeling for example that at least on the larger size Saturn stage it might be more resolved to have the insullation on the inside rather than outside. They didn't like to much \_\_\_\_\_\_ insullation not that I was very happy with it. I had introduced at that time for the very simple defined. For every edge I could get on the mass ratio and I wouldn't have made it and there was no need for in those days specified ainforce and NASA missions there was no long storage time in it. To keep the insulation on \_\_\_\_\_\_\_. So we through it out. It didn't seem to be that terrable a problem and \_\_\_\_\_\_\_ initial problems we solved them and it flies off every time which is rather well . That's not one of the critical areas of the Saturn. Hay look, they had it inside and these sort of changes now the moment you put it inside and you put it against the wall you wouldn't have to fight with the helium layers that we had to have between the steel wall the pressurized structure and the insullation it ficilitated things on the one hand on the other hand it made it more difficult because the huge big structure was breathing it was expanding and contracting and to make sure that this insullation wouldn't crack under those conditions and thereby opened up heat leaks \_\_\_\_\_\_ pretty formitable ones. So they exchanged one problem for another and solved it beautiful. They were in other words they were different. In other words what I'm saying is that they gained from our experience' they could avoid a number of mistakes that we have made and from there they went on to optimize the situation for their own purposes. In the engine field initially nothing was made. To say a word about the engine in the second half of the 50's I worked rather closely with rocket dine I worked with \_\_\_\_\_\_ nuclear engines I worked nuclear and various other large booster engines. I worked with a daring also on a design

Here was to mount on top of this an a reuseable upper stage when people were carried up and a non-reusable traght stage which would be launched automatically. And sent to a space station. I published an article about that at that time In the <u>Journal of American Rocket Society</u>, and for the first time I think tried to get to put economy in this whole thing. We have a standard which either serves maned for possible resuable flight or unmanned for cargo flights. and that assisted very simple cheap cargo bath that maybe doesn't have be returned and so we get more payload into this one. and in all these respects I kept on earging rocketdine to really look seriously in oxygen hydrogen engines. Rocketdine was very strongly on airforce engines they were incompetition with Lockheed on the agenia and they wanted to come up with something that the airforce might like and the airforce was not to interested initially enough in hydrogen they were more interested florine-hydrogene and 204 \_\_\_\_\_\_hydrogen in which they were to have chosen 16

and so yeah we couldn't get them I couldn't really interest them too much and then when centour all of a sudden sold to \_\_\_\_\_\_ that really stung rocketdine they really went into the act and started looking at engines and looking at larger engines. and with the intent of knocking \_\_\_\_\_ Whitney if at all possible out of the satellite they were the saturn five and offer the Marshall team a better engine than \_\_\_\_\_\_\_ six which wasn't the nicest thing in the world but it was already a big forward improvement over any of them. And they were successful with this and they elected a J-2. That is the next biggest engine that the way that the engine was alternately changed and then the saturn one B arrived. And the upper Sentour Slast stage I think it was a C-5 stage or something like that yeah never came about because Saturn one went saturn -B became Saturn one and then of course we had Saturn - . That changes the mission they said it was and not only ? for deep space probe and then they changed their minds and then they wanted to go to the moon. Thats right. And the mission of couse in Saturn-5/became that principle in which we but what had happenedwas that the road was indeed open to oxygen hydrogen and it is now firmly in our arsenal and it would be unthinkable if we really continued to develop another engine to do that if we didn't have that kind of oxygen hydrogen rexperience that we now have through the Saturn

vehicles and through the centour. That how this came about and thats how we often went different paths in the basic approach to the vehicle design but it was based not out of \_\_\_\_\_\_ convictions but simply because we had no alternative with the small like that. I would not have made that design if I had been allowed to build it. \_\_\_\_\_\_ spaceous like the S-2 and the S14 not at all. So after that one was settled we looked in later on at still Aager systems Marshall got interested and wanted to know could we build larger systems and of course, immediately one thing that comes to mind is if you do still larger systems you better be totally reuseable and that led to I got on a contract that led to the ? which which was the first of the one body-type booster vehicles so it was basically blunt and designed for blunt reentry and we had really what looked like a second or third artillery but really just huge payloads. Which gave it then a somewhat more \_\_\_\_\_\_ form but once the payload stages were taken off there was a blunt body that came back into the atmosphere . Dr. Sladon pick that one up fill turn it around came in engine I came in head first that lead to their robust design which followed. But \_\_\_\_\_ and idea of the \_\_\_\_\_ body have become great I know it came out of our stable and persue something like like a blunt vehicle will come about once we get over the shadows and get back right now you can talk to anybody if everybody is \_\_\_\_\_\_ a large frater which gives us to the odd of \$10.00 a pound which the next Columbus design can do. And I think in the 90's we'll return to that thing. And that will be in a way a continuation of the history from the V-2 to the Atlas and the centile and the Saturn. This one You said a space shuttle is kind of a devience from that program? And that program is a

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deviance in the sence that the space shuttle is for small payloads. Small by standards of these large boosters and its a perfectly appropriate payload for the present time 40 thousand pounds or there abouts is a fine payload -I don't care if it's 30 or 60 or 50 but you see when we get to build larger structures and I could hardly begin to tell you why you should build and I think why we will build them by the year 2000 2010structure like these of 2000; 3000-man stopover station and you could not possibly do that with a 40,000 pound launch vehicle. You need launch vehicles that have upwards of a million between a million and pounds or four million pounds. And that are reusable and that is possible in a way I am sorry that I'll be long retired by the time by which this swings back. Because we did a lot of \_\_\_\_\_ and I respect that if I'm still alive to see some of these ideas re-emerge but there is no question that the first step is the shuttle. I should be done but then maybe after we've \_\_\_\_\_ by the late 80's perhaps freighten we'll be alowed to go into a big frater. So now the as I pointed out I always look at where else can I use it is this a dead end street and does this have a growth potential in its development. To me it was not a dead end street. Everthing like hydrocene and long development you have a \_ and then you find out what the next step And so the hydrogen was not only good as a booster but its also a very interest-

wide very intensely at large vehicle wise interplanetary USE . And those have been in study for NASA can be lead to what is somewhat today known as the typical nerva type interplanetary vehicle slone tanks and so on and so forth. Is there a very active nerve program now? The nerve program is not to active at the present . I have testified bfore congress in 59 I was asked to testify and I supported the never program all out there in fact Harry Finger, Stan Ulum, from Los Alomos, Harry Finger from NASA, and some other people were there and also Dale Zish from Eljet in these hearings and my support was based on the assumption which I published back in 59 or 60 somewhere around there that if we do have a nuclear powered vehicle rather than all this chemical stuff wheather heavy or lighter propellent but a nuclear powered vehicle that gives us at least seven or <u>light</u> then we can conduct fast to Mars and to Venus and pointed out the fast flight profiles which would make a quick flight out stay for a little while and then come a longer flight, and then go inside the earth orbit and so on and so forth. I stumbled on that kind of profile originally when I worked for a Science Fiction Novel where I introduced a mishap at the planet and had to return they said would they return? They can't return on the route they have wait 500 days at Mars you see and they're looking at this \_\_\_\_\_\_ all of a sudden there was a whole family of life paths opening itself up that could eventually be used. And from there I went into the utilization solar energy from a nubust near the \_\_\_\_\_ where I not bearing solar distance, solar height solar heat intensity and used a collector and heater in luw he of the reactor to heat my hydrogen and dump it out at 758 hundred per seconds

specific impluse which now was a limit branch and this turned out to be very effective. Although not quite as effective as a Lunas flyby. Which because that's purely gravitation but very much better than striking the earth without a conventional category compulsion system. I almost cut the departure weight of 60 percent apart by this manuver alone. Can you tell this us what the science fiction was that you worked on this? It was never published it was called Expidition Airrace. It was unfortunately never published because I spent more time but I would be glad to give you a copy of it. I still have that thing. I would love to see a copy. O.K. that's absolutely. I send you a copy I have it in my file at home. I'll write you note. Please do I'll mail it to you its all typed out it has a fiction part and it has a technical appendix. But I spent more working on the technical appendix than the fiction part and then I sent it to some science fiction journalist today I consider I would consider I was crazy to try to publish it. It was really well it had strong imphasis on the technology part. But it was then it was in the early 60's that I thought my God we have a whole family of fast \_\_\_\_\_\_ and nobody atall. Von Braun still used very nicely the good old flight rules and everybody used it I used it too. Yeah, but only in my case I wanted to have something going wrong. I dealed conflicts of course in my story and all that and the girl and everything was involved naturally I thought thats necessary to get the science fiction. But anyone would have a mishap and I watched most of my oxygen. In other words what happened was there was a very bad manuver that had to be conducted because they almost collided with an unknown little marsman. I

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introduced that. That hadn't been seen from earth \_\_\_\_\_\_\_ they made a very rapid manuver as a result of this they lost most of the oxygen and their food and it was perfectly clear that if they had to wait 500 days they would all die. And so they started figuring out what can we do how can we go back. In other words it was the first time \_\_\_\_\_\_ anybody thought of and interplanetary mission aboard what do you do with it? In a mission like that. And so I looked at an officer and I saw a number of orbits as so I started calculating. So I was so absorbed in developing more these families of orbits and then published this in the \_\_\_\_\_\_\_. Incedently it wasn't 59 the original novel was written somewhere around 51 or 52 but then over the years I developed more classes and then I published the whole thing in the <u>ASME Journal</u>. In two parts \_\_\_\_\_\_\_

to Venus and Mars introduced a concept that we should have probes with us which should stay in orbit if possible and just land the probes down there and so on and so forth. It how did I need to try to find my way back here. Oh, yeah the idea was that this might be worthwhile if we can do it by  $72^{2}7^{74}$  somewhere around that time. In 59, '60 that looked like a long time away it looked like we might be able to do it. And I had actually I sent you a copy of this ASME thing too I was primarily thinking about either 73 or the 75 mission. I wasn't trying to be too optimistic although I couldn't help myself feeling that maybe we might even make'71. How rediculous that sounds today. About 59 that didn't seem to be so but I was still cautious but I said maybe'75 but I thought I was pretty durned concervative. Well anyway so I supported this, so the years went by in the same dragged on and dragged on. And then they sold President

Kennedy on it. And President Kennedy gave it a boost and it became apparent to me that my God this thing is getting completely out of hand. If you plan a certain thing and see I wanted to started development A here because I wanted to fly Mission A Prime here. Because I wanted to be able to fly mission B over here now all of a sudden A slips down that much so that A-prime to B then I got to refexamine the situation and say do I still want to do this? Should I not wind up in a Block A now and better go to what I really need for B. So I started to look at the gas corriator and then I looked at the pulse the nuclear pulse; and I looked at the fusion. I worked with a number of people in this field including IAC people. And I became convinced that the pulse is the answer. And the pulse was at that time ah brought on at atomics general, general atomics. And general atomics was a sister division of ours so I worked with them and their design is wonderful but I was seeking a bigger design because I'm think of that we do need deffinately to eventually mine another and the only propulsions system that becomes cheaper and of increasing performance the \_\_\_\_. Is the pulse system so that was for me the argument answer I published for years all sorts of things about coagulated the pulse system. Anyway by 1966 my thoughts had coacculated to what I thought was hallucinary I seen this thing dragging along. I talked to Harry Finger and Harry says Oh, what is it? I said I'm gonna stop talking about other propulsions,

if the rover program is being cancelled and won't be replaced later on we won't have nothing so lets at least have that. I says but Harry this thing is turning

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from a driver into a brake. It just holds up everything else. I ean no longer believeng that even those fast reconcernce missions are really sensible. The reason why I'm saying this is it can be very criticle I was in the early years very much deeply out on the probe. And tried to push the probe we got to get some of these programs but primarily I guess initially I really only thought of fly by probes and maybe orbiters and so forth. It became very quickly apparent you know with all the ingenius work that was being done that these probes were capable of a heck of a lot more. And now when you start them sitting down and think about what these folks can do you find very quickly that you are beginning aconnouso to out out compete compete yourself out of the fast manned reconocinse mission. And in one of the last studies for Marshall and NASA where I made some of the advanced studies section at convia back in'63, '64 alright, my conclusion was that number one fly by mission a totally out which we had originally for ment. Totally out because they could be by that time already obviously done so much better by unmanned probes that it made no sense. Fly by mission I figured that to about 21 billion dollars and even something like a voyager was no more than 3 or 4 or 5 billion dollars so in bids per dollar or in useful bids per dollar whatever all that fancy stuff you want to put in on this it became perfactly clear that this is not competable. But many people especially of the younger group members a little bit like Harry Ropper for example they were very capalle of flyby to \_\_\_\_\_ they thought that would be the thing to do and they felt that was 7 North American which felt the only thing could be done. I think that if you soup up the Apollo a little bit then you can indeed make a manned fly-

by to Mars swing up into the \_\_\_\_\_ and then come back. Well, how you can seriously assume that you can do this and penetrate into the totally unknown environment of the Astroid Belt with it possible meteoroid dangers is actually beyond me. Now that was pushed in those days in fact I began to doubt wheather brief capture periods of still use in themselves the basic idea of being that now we may have orbiters long before we know that's right. So what's left is really a type of mission that is really superior to the young man and the kind of mission that one may begin to have serious doubts wheather another can do it. And there I went through a duel evolution evolution about 66 I reached the conclusion that I felt that the nerva wasn't very much worthwhile. We should maybe drop it and directly go to something much better. Preferably the pulse an the Jusion or at least the gas coriator and have first a big time period of unmanned probes. In other words lets put the 70's and the 80's in the unmanned probes and then maybe perhaps in the late 80's or middle 80's middle 80's of course that was 66 so it sounded like we then went and everybody \_\_\_\_\_\_ which we make by 77 we can run the fly by mission. So when you said 82, 84 it sounded like power. 7 today that's no longer of concervative but you know everything changes so fast. But in 66 and then in the 90's have at least the gas ? not the pulse ready. And then we'll go all out really make the . In the meantime I have studied further and have come up with a mission profile which is very exciting could be used by any propulsion system but it does the nerva \_\_\_\_\_ the most good. I just presented that not too long ago to Marshall. There is some in there on the side of the advanced lunar

and banetary propulsion \_\_\_\_\_. And that simply in principle is you take the nerva propelled spacecraft, send the crew out and do nothing to the return capabilities just do \_\_\_\_\_ the return capabilities and then from but but then actually what amounts to practically a space station over there a very very aumptuous sumuous propulsion because your payload is almost increased by a factor of 3 so you have plenty of payload. Then start rotating quotes keep that stuff down with a minimum mission module start making some of the fast flights. That will reduce your stay time to just a few days. I don't want go necessarily to a fly by sort of thing but only reduce it to days which already cuts down the difficulties in the return flight very much compared to the original concept where you want it to stay as long as possible. Maybe six weeks or eight weeks and fly by got closer and closer to the sun and then the then finally window was closing you took the last bus after six weeks stay time was where you really had a hack of a fix you had to rush home as fast as you could. Now you wouldn't do that you would stay only a few days \_\_\_\_\_ the return flights become lot easier. So that's and interesting thing to make maximum use of espeically of the shuttle and of the \_\_\_\_\_. But its still only a preliminary answer. The main line should really be the pulse or something of that sort. But anyway the are still some possibilities open and we may see some of the hydrogen activities derived basically from what originally we all thought I think was a booster possibility. The rover program especially here at North American Rocketdine they were looking strongly at having a nuclear booster. I worker with them at that time so I know about their plans. And because I worked for their booster and they have gotten into the engine the nuclear engine. So the nerva program drags on and drags on and drags on. It still is not without redeeming features. The God if we don't have it at least by the early 80's then I think we should really take the knife to it and but replace it by something else thats good. What is a different story I think we should try everything possible to replace it by a nuclear pulse. So I don't it depends a little bit on what you want wheather you should keep the nerva alive or not. There are alternatives in the \_\_\_\_\_\_ where you don't necessarily have to have a nuclear farry \_\_\_\_\_\_. A nuclear farry has big advantage that its booster which boost away from earth \_\_\_\_\_\_\_ ?

back to earth and that has to be slowed down has to be well \_\_\_\_\_\_ for a rendez vous rondavue at the target and then has to be slowed down and returned before if can be used. You see if you had a duel fly by where you have an elipical orbit and put a space station in elipical orbit and here is the inner termal and this is the outer termal. That elipitcal will osolate between inner and outer terminal you would have a duel fly by situation right? And now you use small fighter-type transports ITD's into orbital interstation transfer league between that elipitcal space station, shuttle station and the terminal space station here and here and a minimum of dead weight you make an appachie transition out to the terminal or from the terminal back into the elipitcal and it goes through the appachie you see. And the same thing on the other end and if you do that then you can with demical propellar come down to a crash factor with which is as good as the classical farry with nuclear. Then you are about 67 pounds of

propellent per pound per payload transported \_\_\_\_\_ trip. From lower to or lower to lunar orbit this is rather similar. So there are various alternatives and if we zeroed in on some of these alternatives then the nerva is going to be in real serious trouble. Am I deviating too much? Put me back on the track. I really interested in what you are talking about unfortunately <sup>7</sup>. Saturn ... I know are getting off the Saturn thing a little bit but so ... you've so much bowled me over with this presentation I just have some minor questions to ask you on Saturn. I'd just as soon have you do whatever you like to do. You've really been wonderful. Now let me just that because our developing because we were I hate to use that word but if you take it judiciously as I mean it we're are the prenciple antagnist of the other line to Marshall I mean the low weight pressure stabilizer under the influence primarily of the genious of Charlie Barset and the successes that we had and the pride of our management even as such as when we broke the centour ice and so forth that our proposal were of course taken looked at most judiciously by our friends at Marshall and we bit for the second stage we bid for the third stage we even bid for the first stage of course we couldn't swing it. We didn't make it I guess some of our in granned \_\_\_\_\_ I was not directly involved in the making beacuse I was busy with centeur at that time but then advanced \_\_\_\_\_\_ made some of the proposals and they were largely relying on pressurized structure. They had some good points but you know nothing is totally and adsolutely rational. People innate have certain inate preferrences and Marshall just didn't feel comfortable with a huge pressurized structure. I can't say wheather rightly or wrongly but I

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certainly can't blame them. I would approace huge big rockets approach the thought of making them a pressurized structure; with much more care that smaller ones this. Maybe it was a beautiful solution for Atlas and maybe it was really not a good solution for Saturn 5 in any case Marshall thought that way and certainly if the other solution was any good we may never know. Certainly the solution they chose was a good one and it works so thats the situation but that got us pretty much out of \_\_\_\_\_. We didn't win one single contract . Back in 61, 62 we didn't make many propsals but I guess we didn't really do what they thought might be the best thing and then they finally did decide that we better design most of it ourselves and \_\_\_\_\_\_ the had. I want not out invents me I want somebody who is doing what I tell them. Boeing seemed to be the one. Yeah, but that was the relationship between Marshall and Boeing butdidn't the S-2/B come up with their own design? In North America? Well, Did the Structures at Marshall have very much to say about the structure design. Well they guided it rather strongly. Now I am not the best guy to talk about details and the origin NAA . 150 book of the S-b because I wasn't with this company at that time. I talked to Bill Parker about it and he seemed to think that it was mostly North American's design and they weren't willing to say that Marshall had very many influences except for the management point of view, visibility and coming out and checking on it but I thats not what I've heard from others. Well I intustively knowing the frame of mind of Marshall not just Marshall and not just the first stage of those days I would be amazed that this is really totally factual I think they had something to say. I must however say that the German team especially those who stayed On the ph

for the Germans had no until 62 or there abouts nothing at all but oxygen procene that was about it. Of course that was it. Well, no question about it. The oxygen-hydrogen when I say rocket just wasn't born at Marshall. And you know industry is perhaps a little bit because of business pressure is on them little bit more versible once we begin to be successful you know rocketdine made that turnabout you know I told you about it ther isn't that much of a tradition \_\_\_\_\_ So I think it is true that probably both Dr's in North America had phycological and somewhat creative freedom. Because the main pride of rocket esteme was deputela deffinately connected with the first stage. With which they were most familiar. had probably a little bit more freedom that Boeing had comparatively speaking. But the mode Marshall in those days was we better won the show. It means that we don't what to get anything that think was wrong . So they keep a tight range on all stage manufacturers but I wasn't here and I cannot say I only do it by influence. Of course all of a sudden Marshall was swamped with the responsibility of three new stages. One stage capturing there attention tremendously the first one. It is possible that they simply couldn't handle it all default also by defalt like a number of decisions go but probably they reviewed them very very carefully. Just one minor point the \_\_\_\_\_\_ insultation used on the first stage is exactly what North American did before they Yeah, that's right. And this is a I can't proove it but I am quite sure because we there was no design that Machine before that whatever there was no design period.

There was absolutely the \_\_\_\_\_\_ work on an engine with the work of the pump \_\_\_\_\_\_\_ and initially I didn't even know anything about the pump. Because when they finally did degide in the Airforce to at least move in the direction of being serious with an hydrogen powered bomb which was \_\_\_\_\_\_\_ they were most afraid of pump. Frankly, I was more afraid of the tanks that of the pumps. \_\_\_\_\_\_\_\_ because I was awol. But the Airforce was afraid of the pump so they said this is maybe the pacesetter so they added \_\_\_\_\_\_\_ and that pump was laid out for feeding the engines of a large bumper but even a huge bumper \_\_\_\_\_\_\_\_ like there is a small rocket. But low and behold I came along with my little centour those people in the know at \_\_\_\_\_\_\_ knew that this